# Catastrophic Interface Debonding in Energetic Materials

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### Solid Propellant Rocket Burning Rate – Chamber Pressure



NASA space shuttle

Solid propellant rocket

• The burning rate increases with the chamber pressure, and the chamber pressure increases with the burning rate.

# **Microstructure of Solid Propellant**







Microstructure of solid propellant material Ide *et al.*, 1999

- Particle/binder with interfaces
- Bimodal particle size distribution
- Large particle volume fraction
- Fracture mainly along interfaces

#### **Crack Propagation along Particle/Binder Interfaces**



• Cracks propagate mainly through interface debonding.

### **Pressure – Burn – Microstructure**



• Relatively small defects, like debonded interfaces and cracks, can lead to catastrophic failure.

### **Objective**

#### To establish a **stability criterion**

for interface debonding in plastic bonded energetic materials

### **What Do Simulations Tell Us?**



Inglis, Geubelle, Matous, Tan and Huang, 2007, Mech. Mater.

- Macroscopic strain imposed as a body force applied at microscale
- 4-noded cohesive element at interfaces
- Periodic boundary conditions

### **Material Subject to Increased Loading**

- Material subjects to equibiaxial strain
- Color-scale represents von Mises stress



### **Catastrophic Interface Debonding**

#### Increasing equibiaxial load



Uniform debonding

Sudden non-uniform debonding

Collapse of interface debonding

Pa

2.00e+04

1 000+04

 $\rightarrow$  Crack formation

Sudden interface debonding under quasi-static loading

### **Deflagration to Detonation Transition**

Increasing chamber pressure



- Sudden interface debonding -> hot spots
- Pressurised cracks

# Strain Energy Density in Hydrostatic Tension



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 $\rightarrow \overline{\varepsilon}$  displacement jump [*u*] at interfaces.

• Determine [**u**] by minimizing Π([**u**])

Tan *et al.*, 2005. *AIAA* Tan *et al.*, 2006. *Int. J. Multiscale Comput. Eng.* 

### **Homogenization Methods in Micromechanics**



• Mori-Tanaka method is extended to account for nonlinear interface debonding.

Tan *et al.*, 2005. *Int. J. Plasticity* Tan *et al.*, 2007. *Int. J. Fract.* 

## Determine Interface Cohesive Law for High Explosives

$$\Pi = \frac{1}{\Omega} \left\{ \frac{1}{2} \int_{\Omega} \sigma_{ij} \varepsilon_{ij} dV + \int_{S_{int}} \phi dA \right\}$$

#### Macroscopic cohesive law

#### PBX 9501 Specimen CCD Camera When $\Delta = 0.53$ mm When ∆=0.65mm (x10<sup>-3</sup>) When ∆ = 0.80mm Cohesive stress, $\sigma/E$ 0.5 -0.5 0.05 0.1 0.15 0.2 Opening displacement, $\delta$ (mm)

#### **Interface cohesive law**

softening modulus: interface strength:  $\widetilde{k}_{\sigma} = 17MPa / mm$  $\sigma_{\rm max} = 1.66MPa$ 



### **Interface Displacement Jumps**

Goal: 
$$\Pi([\mathbf{u}]) = \frac{1}{\Omega} \left\{ \frac{1}{2} \int_{\Omega} \sigma_{ij} ([\mathbf{u}]) \varepsilon_{ij} ([\mathbf{u}]) dV + \int_{S_{int}} \phi([\mathbf{u}]) dA \right\}$$



Interface opening [*u*]

$$\frac{[u]}{a} = A_0 + \begin{cases} A_u \cos \theta & \theta \in [0, \pi/2] \\ 0 & \theta \in [\pi/2, \pi] \end{cases}$$

#### **Quadratic Form of Total Potential Energy**

• Total potential energy density can be expressed in a **quadratic** form of *A<sub>u</sub>* when subject to hydrostatic loading

$$\Pi(A_u) = c_0 + c_u A_u + c_{uu} A_u^2$$

For rigid particle embedded in incompressible matrix under hydrostatic tension:

$$c_{uu} = \frac{3}{2} \mu_m \left( 1.0225 - f - \frac{\tilde{k}_\sigma a}{6\mu_m} \right)$$

- f: particle volume fraction
- $\widetilde{k}_{\sigma}$ : interface softening modulus
- *a*: particle radius
- $\mu_m$ : shear modulus of matrix

#### **Stability Criterion for Interface Debonding**



A<sub>u</sub>: Magnitude of non-uniform interface opening

### **Case Study I: Particle Size Effect**



#### **Plastic Bonded Explosives**

interface softening modulus:	$\widetilde{k}_{\sigma} = 0.017 MPa / \mu m$	Tan <i>et al.</i> , 2005
large particle volume fraction:	f = 69.5%	Skidmore et al., 1997
matrix modulus:	$\mu_m = 0.334 MPa$	Cady et al., 2000

 $a_{cr}^{stability} \approx 39 \mu m$ 



# **Performance versus Sensibility**

 $a_{cr}^{stability} \approx 39 \mu m$ 



Balancing the performance and sensibility through changing the particle size distribution

### **Size Effect on DDT**



→ Deflagration to Detonation Transition (DDT)

## Case Study II: Macroscopic Deformation during Compression

8% volume increase when HMX particles transfer from  $\beta$ -phase to  $\delta$ -phase. PBX 9501 High Explosive





Average radius of coarse particles is 125µm.

- Microscopically, each particle may debond suddenly in a random direction.
- Macroscopically, deformation field is chaotic.

## **Experimental Setup for Slow Heating of PBX 9501 Sample**



- Cubic sample of size 12.7mm
- Free standing sample
- Temperature ramps from 33°C to 195°C in one hour

### **Evolution of the Deformation Field**



### **Strain Field**



below transformation temperature



#### around transformation temperature

• low strain -> chaotic deformation field

# Summary

• A stability criterion for interface debonding is established for energetic composite materials.

• Catastrophic interface debonding is observed in numeric simulations.

 Catastrophic interface debonding contributes to the chaotic deformation during the phase transformation of HMX particles.

• Application: safety of solid rocket propellant.







